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Buckling of pressurized spherical shells containing a through-thickness defect



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ABSTRACT

We present a study on the pressure buckling of thin, elastic spherical shells containing a thickness defect. Methodologically, we combine precision model experiments, finite element simulations, and a reduced axisymmetric shell model. We observe qualitatively different buckling behavior by varying the geometry of the defect: either one buckling event or two events comprising local buckling at the defect and global buckling of the entire shell. We systematically analyze the loading path for the imperfect shell under prescribed pressure or volume change and identify three buckling regimes. We then explore a wide parameter space to study the dependence of the buckling regimes on the defect geometry, thus obtaining a phase diagram with quantitative relationships between critical buckling pressures and defect geometry. We find that the global buckling becomes insensitive to the defect beyond a critical value of its amplitude, and we demonstrate that the buckling regimes are governed by the three geometric parameters of the defect, namely its width, amplitude and the width of the transition region across the edge of the defect.

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1. Introduction

Shell structures are observed regularly in nature and employed routinely in technology, across length scales, with the buckling of shells often playing an important role in biology (Katifori et al., 2010; Lidmar et al., 2003; Yin et al., 2008) and in the design of engineering structures (Hilburger, 2012; Pedersen and Jensen, 1995). From a traditional view point, the catastrophic collapse caused by buckling is seen as the first route to the failure of structures, thus large efforts have been made to understand shell buckling and provide criteria or predictions for the critical buckling load for structural design (Babcock, 1983; Hilburger et al., 2006; Samuelson and Eggwertz, 1992). In the recent decades, the large deformation accompanying buckling has been seen as an opportunity for shape morphing in the design of novel materials and structures with new functionalities (Reis, 2015; Shim et al., 2012; Wang et al., 2014; Xu et al., 2015). This new perspective has stimulated further research on the fundamental aspect of buckling, with a particular attention to imperfection sensitivity and probing forces (Baumgarten and Kierfeld, 2018; 2019; Gerasimidis et al., 2018; Haynie et al., 2012; Hutchinson, 2016; Hutchinson and Thompson, 2017a; 2017b; Knoche and Kierfeld, 2011; 2014; Kriegesmann et al., 2012; Lee et al., 2016b; Marthelot et al., 2017; Paulose and Nelson, 2013; Virot et al., 2017). In these recent studies, cylindrical shells under axial compression and spherical shells under external pressure are two common settings used to understand the underlying mechanism of shell

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buckling and its imperfection sensitivity. In this paper, we will focus on the pressure buckling of spherical shells, for which we now review the relevant literature.

Under uniform pressure loading, the classic prediction for the critical buckling load of a perfect spherical shell with radius R and thickness h_0 is (Zoelly, 1915)

$$p_{\rm c} = \frac{2E}{\sqrt{3(1-\nu^2)}} \left(\frac{R}{h_{\rm o}}\right)^{-2},\tag{1}$$

where E and ν are the Young's modulus and Poisson's ratio of the material, respectively. However, this theoretical prediction always overpredicts the buckling pressure, p_{max} , measured in experiments (Carlson et al., 1967; Kaplan and Fung, 1954; Krenzke and Kiernan, 1963; Tsien, 1942). The ratio of p_{max} to p_{c} defines the knockdown factor of a shell, quantifying the difference between the actual buckling load and the theoretical prediction, as

$$\kappa_{\rm d} = \frac{p_{\rm max}}{p_{\rm c}} \,. \tag{2}$$

From the survey of historical experimental results presented in our previous work (Lee et al., 2016b), knockdown factors are widely distributed and fall in the range between 0.1 and 0.3, for the worst cases. The low value of knockdown factors and their large variability are due to the high imperfection sensitivity of shells, since pressure buckling is a subcritical bifurcation. These imperfections often arise during the fabrication process within the material or the geometry of the shell.

The imperfection-sensitive nature of shell buckling requires the proper consideration of defects in both theory and simulations in order to attain a reliable prediction of the critical buckling load. The first attempts to rationalize the knockdown factor were carried out by von Kármán and Tsien (1939) and Tsien (1942). Subsequently, Koiter (1945) developed a general theory for the stability of elastic systems subject to a conservative loading, in which the imperfection sensitivity was related to the initial post-buckling behavior of the perfect structure. Since then, several studies based on Koiter's theory have been conducted to assess the effect of imperfections of different types, including geometry (Budiansky and Hutchinson, 1972; Hutchinson et al., 1971), loading (Bijlaard and Gallagher, 1960), and boundary conditions (Kobayashi, 1968). Nonetheless, this body of work has led mostly to a qualitative understanding of the intricate relationship between critical load and imperfections. Statistical approaches have also been explored to study the imperfection sensitivity, although this requires the knowledge of defect distributions that cannot be easily obtained (Elishakoff, 2014). The lack of a fabrication technique where defects could be eliminated or at least controlled made it difficult to test theoretical predictions against precision experimental results involving a systematic exploration of the primary parameters. For decades, experiments lagged behind both theory and simulations (Chen, 1959; Hutchinson et al., 1971; Seaman, 1962). This gap between theory and experiments results in a fundamental lack of understanding, which has so far forced the practical design of shells to be based on dated empirical guidelines that largely overestimate the critical loads to guarantee safety.

In engineering practice, with the development of advanced fabrication techniques, imperfections in shell structures have been remarkably reduced. Therefore, the old design guidelines tend to be overly conservative and no longer applicable to the current structural design (Hilburger, 2012). Additionally, buckling is increasingly being used to achieve functionality as in microcapsules (Datta et al., 2012; Vian and Amstad, 2019), metamaterials (Kochmann and Bertoldi, 2017; Rafsanjani et al., 2019), and robots (Jampani et al., 2018; Whitesides, 2018). As such, a thorough understanding of how defects interact with buckling is desirable, so as to be able to tune the buckling conditions on demand.

A breakthrough in experiments was recently made by Lee et al. (2016a) with the development of a rapid prototyping technique to fabricate thin shells that are applicable to precision laboratory research settings. This technique is based on coating a spherical mold with a polymer solution, and can be modified to allow for the precise engineering of a defect in the shell (Lee et al., 2016b; 2019; Marthelot et al., 2017). By modifying the original coating technique, Lee et al. (2016b) introduced a local dimple-like geometric defect to the shell, whose amplitude and width could be varied systematically. The authors then reported the quantitative relationship between the knockdown factor and defect geometry. In parallel, simulations through finite element modeling (FEM) and a numerical analysis based on small-strain and moderate-rotation theory were also carried out, yielding results in good agreement with experiments. This study demonstrated that the knockdown factor can be accurately predicted, provided the imperfections are taken into account and modeled appropriately.

The experimental advances mentioned above have played an important role on revival of the interest on the buckling of both perfect and imperfect shells (Hutchinson, 2016; Hutchinson and Thompson, 2017a; 2017b; 2018). In particular, Hutchinson (2016) studied the load-carrying behavior of imperfect shells subject to uniform external pressure in the pre-buckling and post-buckling regimes, showing a strong dependence of the knockdown factor on the type of the defect. Gerasimidis et al. (2018) performed a systematic study through FEM simulations on the difference between a dimple-like defect embedded in the geometry and a dimple-like defect created by a probing force. They concluded that the perturbation induced by a probing force leads to larger knockdown factors than the equivalent geometric defect. Moreover, a non-destructive probing technique was proposed and successfully applied in experiments to access the landscape of stability of cylindrical shells under axial compression, without any foreseeing on imperfections (Hutchinson and Thompson, 2017b; Thompson, 2015; Thompson and Sieber, 2016; Virot et al., 2017). It is still debatable whether this technique can be used for spherical shells (Marthelot et al., 2017; Thompson et al., 2017).

Whereas several past studies have focused on the effect of defects of the geometry of the middle surface, much less attention has been dedicated to imperfections in the thickness profile. Hutchinson et al. (1971) numerically studied the ef-

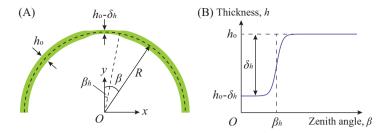


Fig. 1. (A) Schematic diagram of a shell containing a through-thickness defect. (B) Thickness profile as a function of the zenith angle.

fect of a local axisymmetric thickness variation on the buckling of cylindrical shells under axial compression. Experiments on the buckling of non-homogeneous capsules under an external osmotic pressure were carried out by Datta et al. (2012). The capsule thickness was gradually varied from a minimum value at one of its poles to a maximum value at the other pole. Both through experiments and simulations, it was found that the onset of buckling initiates from the thinner region, which, therefore, sets the critical osmotic pressure. In a related study, Paulose and Nelson (2013) numerically investigated the buckling of spherical shells with a thinner region under prescribed pressure loading. The shell exhibited different buckling pathways affected by the soft spot; either a single buckling event or two separate buckling events could be observed. They also analyzed the dependence of buckling transitions on the dimensions of the soft spot through numerical simulations and theoretical arguments. These studies motivate a research effort to fundamentally understand the influence of thickness defect on the buckling of spherical shells via experiments, FEM, and reduced shell theory.

Here, we study the buckling of pressurized spherical shells containing a defect consisting in a localized region of reduced thickness. Our focus is to both quantify and rationalize how this *through-thickness* imperfection modifies the critical buckling conditions compared to the classic prediction of Eq. (1), *i.e.*, the knockdown factor. Qualitatively different buckling behavior depending on the geometry of the defect is observed and analyzed by combining experiments, finite element simulations, and axisymmetric shell theory. We identify three distinct buckling regimes involving local buckling at the defect and global buckling of the entire shell. The quantitative results are summarized in a phase diagram, where the relationships between critical buckling pressures and defect geometry are elucidated.

Our paper is organized as follows. In Section 2, we define our problem, and the experimental, numerical and theoretical methodology followed to investigate it is presented in Section 3. The experimental results of the critical buckling pressures of our fabricated shells are reported in Section 4. In parallel, we perform FEM simulations using the experimentally measured geometric profiles and validate the FEM against experiments. In Section 5, we then use the validated FEM to systematically study the buckling behavior of spherical shells affected by the through-thickness defect in a wider parameter space than that available experimentally. Furthermore, we juxtapose predictions from the axisymmetric shell theory onto the FEM results. In Section 6, we discuss and summarize our findings.

2. Definition of the problem

We consider a thin, elastic, imperfect hemispherical shell of radius, R, and thickness, h_0 , clamped at its equator and loaded under uniform pressure (see schematic in Fig. 1). The shell is imperfect since it contains a through-thickness defect at its pole, where there is a local region of reduced thickness; the thickness is constant elsewhere (h_0) . Let β be the zenith angle with the north pole located at $\beta=0$. The through-thickness defect is located in the region $0 \le \beta \le \beta_h$, where β_h is the half angular width of the defect, and the shell thickness is $h_0-\delta_h$. We shall refer to δ_h as the amplitude of the defect, and h_0 as the nominal thickness of the shell.

To generalize the description of the through-thickness defect, we consider a hyperbolic tangent profile

$$\delta(\beta) = \frac{1}{2} \delta_h \left(1 - \tanh\left(\frac{\beta - \beta_h}{\gamma_h}\right) \right),\tag{3}$$

where γ_h is an angular parameter setting the width of the transition region across the edge of the defect. Consequently, the thickness profile of the imperfect shell is

$$h(\beta) = h_0 - \delta(\beta). \tag{4}$$

Overall, the geometry of the generalized through-thickness defect is characterized by its amplitude δ_h , half angular width β_h , and the transition-region parameter γ_h . We normalize δ_h and $h(\beta)$ by the nominal thickness, h_o , to define the normalized defect amplitude $\bar{\delta}_h = \delta_h/h_o \in [0,1)$ and the normalized thickness profile $\bar{h}(\beta)$. Following Hutchinson (2016), Lee et al. (2016b) and López Jiménez et al. (2017), who showed that the buckling behavior of thin elastic shells is independent of the radius-to-thickness ratio, we scale β_h and γ_h , respectively, as

$$\overline{\beta}_h = \beta_h \left(\sqrt{12(1-\nu^2)} R/h_0 \right)^{\frac{1}{2}},\tag{5}$$

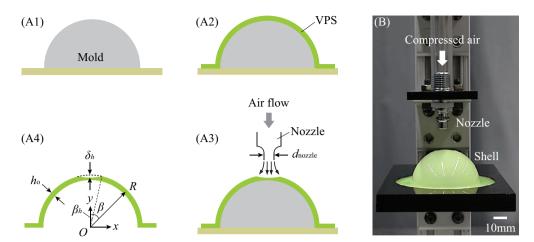


Fig. 2. Fabrication of thin shells containing a through-thickness defect. (A) Schematic diagrams of the experimental protocol. (A1) A metal hemisphere (radius $R = 25.4 \,\mathrm{mm}$) is used as a rigid mold. (A2) The mold is coated with liquid VPS (see text for details). (A3) Immediately after the pouring of the VPS solution, an air jet with a constant flow rate is impinged onto the north pole of the liquid film, thereby generating a steady-state crater under the nozzle. (A4) Upon curing of the VPS film, the thin and elastic shell containing a single through-thickness defect at its north pole is peeled off from the mold. (B) Photograph of the experimental setup used to fabricate thin hemispherical shells containing a through-thickness defect.

and

$$\overline{\gamma}_h = \gamma_h \left(\sqrt{12(1-\nu^2)} R/h_0 \right)^{\frac{1}{2}}. \tag{6}$$

In Section 5.6, we shall show that the buckling behavior of our imperfect shells is entirely governed by the geometric parameters $\overline{\delta}_h$, $\overline{\beta}_h$, and $\overline{\gamma}_h$.

The generalized description of the through-thickness defect presented above corresponds to a thinning of the shell from both its inner and outer surfaces, such that the middle surface of the shell remains spherical. This is the general geometry that will be used in the parametric exploration presented in Section 5 using the FEM and axisymmetric shell theory. On the other hand, in the experiments (Section 3.1.1), we introduce a defect by impinging the outer surface of a shell with an air jet during shell fabrication, while constraining its inner surface to a spherical mold. This protocol yields a shell whose outer surface at the defect is concave, while the inner surface remains spherical. As such, the defect represents an imperfection in both the overall thickness, as well as in the non-spherical shape of the middle surface. Therefore, the FEM simulations shall first be performed using the actual experimentally measured profiles (Section 4) for validation purposes, prior to focusing on the generalized defect profile of Eq. (3) for a more extensive exploration of the parameter space.

3. Methods

We proceed by presenting the experimental, numerical and theoretical methodology. For our precision experiments, we fabricated shells containing a through-thickness defect using a coating technique. In Section 3.1, we describe our fabrication technique, the procedure that we have followed to characterize the geometry of the fabricated shells, and the measurement of their critical buckling pressures. In parallel to the experiments, we also conducted FEM simulations and developed a reduced axisymmetric shell model, which are presented in Sections 3.2 and 3.3, respectively.

3.1. Experiments

3.1.1. Fabrication of the shell specimens

We fabricated our thin, elastic hemispherical shells using a polymer coating technique developed in a previous study (Lee et al., 2016a). The basis of this technique consists of first producing a thin liquid film by the gravity-driven drainage flow of a viscous elastomeric solution, which is poured onto a hemispherical mold. Upon curing, the polymerized solid film yields an elastic shell of nearly uniform thickness. For the current study, we have modified the above protocol to be able to fabricate shells containing a through-thickness defect (with customizable amplitude and width) by inducing a localized variation of the thickness of the polymer film during curing.

In Fig. 2, we present a series of schematic diagrams summarizing our fabrication protocol. We coated the surface of a rigid hemisphere (stainless steel ball, radius $R = 25.4 \,\mathrm{mm}$, TIS-GmbH, Germany) with a vinylpolysiloxane (VPS) polymer solution (Elite Double 32, Zhermack, Italy), which cures into a silicone-based elastomer. The VPS solution was mixed with a ratio of base to catalyst of 1:1 in weight, for 10s at 2000 rpm (clockwise), and then 10s at 2200 rpm (counterclockwise) using a centrifugal mixer (ARE-250, Thinky Corporation, Japan). The solution was then poured onto the mold, 290s after

Table 1Nozzle diameters, flow rates, and the yielded defect amplitudes in fabrication.

Nozzle diameter $d_{ m nozzle}$ [mm]	Flow rate q[L/min]	Defect amplitude $\overline{\delta}_h$
3.5	[0.4, 1.6]	[0.06, 0.83]
4.0	[0.6, 3.0]	[0.18, 0.80]
4.8	[0.5, 4.0]	[0.03, 0.73]

the mixing so as to increase its viscosity to a desired value (Lee et al., 2016a). The gravity-driven viscous flow yielded a thin polymer layer with nearly uniform thickness. Immediately after pouring, a steady jet of compressed air with a constant flow rate coming out of a nozzle (diameter d_{nozzle} , fixed at a height of 15 mm above the mold) was imposed to steadily impinge onto the pole of the polymer film until it cured. Since this film was constrained at its inner surface by the rigid hemispherical mold, a crater was generated on the outer surface of the film as a result of the air impingement. This crater of amplitude δ_h set the thin region in the shell. The drainage of excess polymer formed a thicker band (puddle) at the equator, approximately 3 mm in thickness, which ensures the clamped boundary condition there.

Upon curing (in approximately 20 min at room temperature 22 ± 0.4 °C), the polymer film was peeled from the mold, yielding a thin, elastic hemispherical shell containing a through-thickness defect. The fabricated defect comprises a region in the shell of reduced thickness, but, as we mentioned in Section 2, it also modifies the shape of the middle surface. The produced defect is, therefore, a combination of a 'pure' through-thickness defect and a dimple-like geometric defect. However, as we shall show in Section 5, with this defect, the experimental shells exhibit buckling behavior that is qualitatively similar to shells containing the pure through-thickness defect described by Eq. (3).

To characterize the nominal thickness, h_0 , of the shells, we cut off a narrow strip ($\approx 2 \times 6 \,\mathrm{mm}^2$) from the shell, along a meridian at $\beta = 45^\circ$ and measured the thickness of this strip using a microscope (VHX-5000, Keyence Corporation, Japan). Note that the thickness measurements were performed *after* the buckling pressure measurements (Section 3.1.3), so as to have the precise thickness value for the specimen used for a specific measurement. As demonstrated in our previous work (Lee et al., 2016a), shells with different values of h_0 could have been attained by varying the waiting time between the preparation of the polymer solution and the moment of pouring onto the mold. However, given the otherwise already vast parameter space, in our experiments, we have decided to fix the nominal thickness at $h_0 = 269 \pm 10 \,\mu\mathrm{m}$ (the uncertainty corresponds to the standard deviation of 42 fabricated shells), which translates into a radius-to-thickness ratio of $R/h_0 = 94$.

The fabrication technique developed here also allows us to systematically tune the amplitude of the defect by varying the air flow rate, q, and the width of the defect by varying the nozzle diameter, d_{nozzle} . Impinging the polymer layer at a higher flow rate will produce a deeper crater, and, hence, a larger defect amplitude. We have used three different nozzles with diameters $d_{\text{nozzle}} = \{3.5, 4.0, 4.8\}$ mm. For each nozzle, we adjusted the air flow rate to vary defect amplitude and obtained three sets of shells (Table 1). Next, we detail the characterization of the geometry of these defects.

3.1.2. Characterization of the defect geometry through profilometry

We characterized the geometry of the experimental shells containing a through-thickness defect using an optical profilometer (VR-3200, Keyence Corporation, Japan). After curing but prior to demolding the shell, the 3D profile of its outer surface was measured and averaged latitude-wise to obtain an axisymmetric 2D profile, $Y_0(x)$, with the pole located at $(0, Y_0(0))$. Since, during these measurements, the shell was still adhered to the rigid mold, we assume that its inner surface is a hemisphere of radius R = 25.4 mm. After the experimental campaign of critical pressure was completed for a particular shell (more details in Section 3.1.3), we cut off a small piece at the pole of the shell and also measured the local thickness there, h_p . Then, the centroid of the shell was determined as $O(0, Y_0(0) - h_p - R)$. With this information at hand, we can now transform $Y_0(x)$, the 2D profile of the outer surface, from the (x, y) Cartesian coordinates into the (r, β) polar coordinates with the origin at $O(0, A_0(\beta))$. Naturally, the 2D profile of the inner surface is $R_1(\beta) = R$. Then, the radial distance between the two surfaces, $h(\beta) = R_0(\beta) - R_1(\beta)$, is the thickness profile of the shell. Due to limitations in the field of view of our optical profilometer, we could only measure the profile in the range of approximately $0 \le \beta \le \pi/6$, which can, however, still fully cover all the defects of the experimental shells investigated. Out of this range, the thickness is assumed to transition linearly to h_0 at $\beta = \pi/4$ and then keep a constant of h_0 from $\beta = \pi/4$ to $\pi/2$.

In Fig. 3, we present three sets of representative thickness profiles (normalized by the nominal thickness h_0) of shells fabricated using the three nozzles with increasing diameters $d_{\text{nozzle}} = \{3.5, 4.0, 4.8\}$ mm. All thickness profiles exhibit a minimum near the defect, at $\beta = 0$, and gradually recover to the nominal thickness towards the equator. For each set, we tuned the defect amplitude by adjusting the flow rate of the air jet that impinged the uncured polymer film during the coating process. For example, with $d_{\text{nozzle}} = 3.5$ mm (Fig. 3A), the normalized defect amplitude $\overline{\delta}_h$ is increased from 0.17 to 0.80 by increasing the air flow rate from 0.5 to 1.3 L/min. The defect width also increases with nozzle diameter, a trend that becomes more pronounced for larger defect amplitudes.

In short, by adjusting the air flow rate and nozzle diameter, our experimental fabrication protocol allows us to fabricate shells with a customizable defect amplitude and width, over a wide parameter range. In the experiments, the normalized

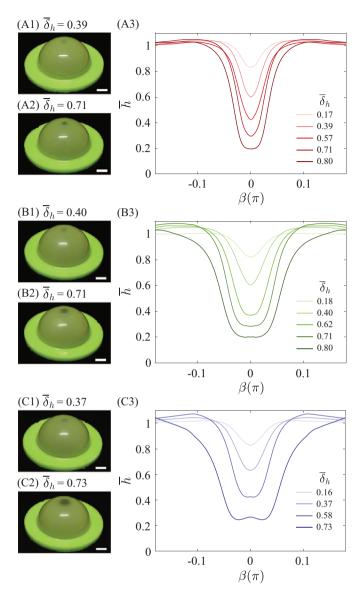


Fig. 3. Photographs and thickness profiles of representative shells containing a through-thickness defect fabricated using air nozzles with different diameters. (A1, A2) $d_{\text{nozzle}} = 3.5$ mm, (B1, B2) $d_{\text{nozzle}} = 4.0$ mm, and (C1, C2) $d_{\text{nozzle}} = 4.8$ mm. The values of the air flow rate for each of the representative shells are: (A1) q = 0.7 L/min, (A2) q = 1.0 L/min, (B1) q = 1.0 L/min, (B2) q = 2.0 L/min, (C1) q = 1.3 L/min, and (C2) q = 4.0 L/min. The corresponding normalized defect amplitudes are: (A1) $\delta_h = 0.39$, (A2) $\delta_h = 0.71$, (B1) $\delta_h = 0.40$, (B2) $\delta_h = 0.71$, (C1) $\delta_h = 0.37$, and (C2) $\delta_h = 0.73$. (A3, B3, C3) Thickness profiles of each plot) are achieved by varying the air flow rate in the ranges (A3) $0.5 \leq q$ [L/min] ≤ 1.3 , (B3) $0.6 \leq q$ [L/min] ≤ 3.0 , and (C3) $0.8 \leq q$ [L/min] ≤ 4.0 . Scale bars in (A)-(C) are 10 mm.

defect amplitude is explored in the range $0.03 \le \overline{\delta}_h \le 0.83$, and the normalized half angular width in the range approximately $1.1 \le \overline{\beta}_h \le 3.3$ ($\pi/50 \le \beta_h \le \pi/16$).

3.1.3. Measurement of the critical buckling pressure

The pressure loading of our shells was performed by imposing the volume change of the air inside the shell. To do so, we mounted the shell onto an acrylic plate containing a through hole at its center, which was connected to both a stepper motor driven syringe pump (NE-1000, New Era Pump Systems, Inc., Farmingdale, NY) and a differential pressure sensor (MPXV7002, NXP Semiconductors, The Netherlands). The system was air-tight due to a combination of the van der Waals adhesive forces between the equatorial rim of the shell and the acrylic plate, as well as the water film that was added at the edge of the rim to further prevent leakage of air through the gap.

For each experimental run, the syringe pump was set to expand the air inside the loading system, which comprises the shell, syringe and connecting tube, at a constant rate of 0.04 mL/min until the shell buckled. Meanwhile, the pressure

differential between the outside (atmospheric pressure) and inside of the shell was monitored by the pressure sensor at an acquisition rate of 1 Hz. For each tested shell, we obtained the evolution of the pressure loading, p, as a function of the volume change imposed by the syringe, Vim. The critical buckling pressure is defined as the peak value of this signal, which represents the load-carrying capacity of the shell. Note that, in the experiments, the imposed volume change of the air inside the loading system, V_{im} , is not exactly the volume change of the pressurized shell, V_{im} , due to the compressibility of air. Given the volume of the air inside our loading system and the stiffness of our shells, Vim for a perfect hemispherical shell is $\approx 1.25V$, in the pre-buckling regime. We estimated this value using the ideal gas law at the initial state, pressure differential p = 0, compared to the pressurized state (the temperature is assumed to be constant during loading). The details for the computation of this prefactor of 1.25, which is specific to our loading system, are provided in Appendix A. Past the onset of buckling, the feedback between the internal pressure and volume will make the volume change of the shell unable to be strictly prescribed by this system, thus, will affect the bifurcation behavior in the post-buckling regime (Knoche and Kierfeld, 2014). In any case, the primary quantity that is of interest in the experiments is the critical buckling pressure (rather than V), which can be directly and accurately measured from the pressure sensor.

3.2. Finite element modeling

FEM simulations were performed using the commercial finite element modeling package Abaqus/Standard. The hemispherical shell was reduced to a 2D axisymmetric model. A through-thickness defect with either the geometry measured in experiments or the generalized geometry described in Eq. (3) was imported into the model, as will be detailed later. The elastomer was modeled as an incompressible neo-Hookean material with a Young's modulus $E=1.16\pm0.03\,\mathrm{MPa}$, which was experimentally measured through dog-bone tests. The Poisson's ratio was assumed to be 0.5. The shell was discretized using reduced hybrid axisymmetric solid elements (CAX4RH) with a regular mesh. The mesh had a constant of 8 elements in the thickness direction, and the element size in the longitudinal direction was varied according to the thickness profile in order to generate elements with an aspect ratio of approximately 1. A mesh convergence study was performed to ensure that the mesh was sufficiently fine. Geometric nonlinearities were taken into consideration in the simulations.

In the finite element analysis using the experimentally measured geometry, we imported the 2D profile of the shell outer surface measured after demolding, $r_0(\beta)$, through optical profilometry. This way, we can take into account any slight changes in the geometry of the shell away from a perfect hemisphere imparted by the fabrication process (e.g., due to gravity or the pre-stresses developed in the constrained polymer layer during curing), for a direct comparison between the experimental and numerical results. Eventually, $r_1(\beta) = r_0(\beta) - h(\beta)$ and $r_0(\beta)$, the experimentally obtained profiles for the inner and outer geometry of the shell, respectively, were used to determine the shell geometry. In the FEM simulations using the generalized geometry, the through-thickness defect involves thinning a perfect shell from both its inner and outer surfaces, by the same amount, as was stated in Section 2. As such, the 2D profiles of the inner and outer surfaces of an imperfect shell can, respectively, be written as

$$r_{i}(\beta) = R - (h_{0} - \delta(\beta))/2 \tag{7}$$

and

$$r_0(\beta) = R + (h_0 - \delta(\beta))/2. \tag{8}$$

both of which determine the shell geometry.

The shell was loaded by a uniform live pressure applied on its inner surface, and clamped at the equator. The simulations were conducted using the Riks method, which solves for equilibrium states by prescribing the arc-length of loaddisplacement curve, allowing us to follow both stable and unstable equilibrium paths. For each simulation, we obtained the following quantities:

- (i) $\overline{p} = p/p_c$, the pressure loading normalized by the classic buckling pressure p_c ;
- (ii) $\overline{V} = V/V_c$, the volume change normalized by the volume change of a perfect hemispherical shell immediately prior to buckling at \overline{p}_c , that is $V_c = 2\pi (1 - \nu) R^2 h_0 / \sqrt{3(1 - \nu^2)}$ (Hutchinson and Thompson, 2017a); (iii) $\overline{w} = w/h_0$, displacement at the pole normalized by the nominal thickness of the shell h_0 .

Throughout, we assume axisymmetry, both in the pre-buckling and post-buckling configurations. This choice is motivated by the fact that, for a perfect shell with $R/h_0 = 100$, non-axisymmetric bifurcation can be found only when the pole deflection reaches $w/h_0 = 22$, deep into the post-buckling range (Hutchinson and Thompson, 2017a). Also, for shells with an axisymmetric dimple imperfection, non-axisymmetric bifurcations do not occur prior to attainment of the maximum pressure in the axisymmetric state (Hutchinson, 2016). However, specifically for shells containing a through-thickness defect presented in this study, a large defect that produces an excessively wide or/and thin region could break the axisymmetry of buckling (Paulose and Nelson, 2013), which is beyond the scope of this study. Thus, we carefully examined the post-buckling behavior of representative shells within the explored parameter space by full 3D FEM. For the results presented in this paper, no non-axisymmetric bifurcations were found in the vicinity of the buckling onset, but we shall indicate the area in the parameter space where the break of axisymmetry occurs.

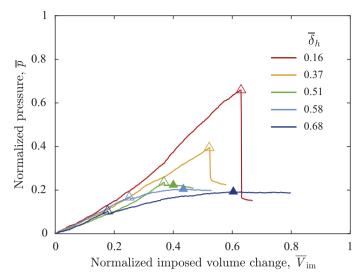


Fig. 4. Typical load-carrying curves, showing the normalized pressure, \overline{p} , as a function of the normalized volume change of the air inside the loading system imposed by the syringe, $\overline{V}_{\rm im}$. The critical buckling pressures associated with each curve are represented by the corresponding large symbols, open for \overline{p}_1 and closed for \overline{p}_2 . The defects were fabricated using the large nozzle $d_{\rm nozzle} = 4.8$ mm. The normalized defect amplitude was varied $\overline{\delta}_h = \{0.16, 0.37, 0.51, 0.58, 0.68\}$. The various values of $\overline{\delta}_h$ are detailed in the legend.

3.3. Theoretical description of axisymmetric (1D) shells

Within the framework of small strains and large displacements/rotations, we use a shell model whose implementation has been recently reported in Pezzulla and Reis (2019). Next, we provide a brief outline that summarizes this recent work. First, the dimensionless elastic energy of the shell is written as

$$\overline{\mathcal{U}} = \int_{0}^{\pi/2} h(s) [(1 - \nu) \operatorname{tr} (\mathbf{a} - \mathring{\mathbf{a}})^{2} + \nu \operatorname{tr}^{2} (\mathbf{a} - \mathring{\mathbf{a}})] R^{2} \sin s \, ds
+ \frac{1}{3} \int_{0}^{\pi/2} h(s)^{3} [(1 - \nu) \operatorname{tr} (\mathbf{b} - \mathring{\mathbf{b}})^{2} + \nu \operatorname{tr}^{2} (\mathbf{b} - \mathring{\mathbf{b}})] R^{2} \sin s \, ds,$$
(9)

where **a** and **b** are the metric and curvature tensors of the deformed middle surface of the shell, whereas $\overset{\circ}{\bf a}$ and $\overset{\circ}{\bf b}$ are those related to the undeformed middle surface (Niordson, 1985). The trace operator in the metric $\overset{\circ}{\bf a}$ is denoted by 'tr', and the s coordinate runs along meridians. The energy functional in Eq. (9) has been normalized by $\pi E/(4(1-\nu^2))$. Parametrizing the deformed profile curve of the middle surface as ${\bf r}(s)=(\phi(s),\psi(s))$, where ϕ and ψ are the 2D Cartesian coordinates, the potential associated to the live pressure ${\cal P}=pV$, itself normalized by $\pi E/(4(1-\nu^2))$, can be written as (Pezzulla and Reis, 2019)

$$\overline{P} = \frac{8}{3} (1 - v^2) \frac{p}{E} \int_0^{\pi/2} \phi(\psi \phi_s - \psi_s \phi) \, ds \,. \tag{10}$$

Equilibrium equations can then be generated by imposing $\delta(\overline{\mathcal{U}}+\overline{\mathcal{P}})=0$, for all possible variations $\delta\phi$ and $\delta\psi$. Regarding the displacement at the north pole as the loading parameter since shell buckling is subcritical, the corresponding equilibrium solutions are computed via the Newton-Raphson method.

4. Buckling of the fabricated shells: Experiments versus FEM

In this section, we investigate the load-carrying behavior and critical buckling pressures of the three sets of shells fabricated using the nozzles with $d_{\text{nozzle}} = \{3.5, 4.0, 4.8\}$ mm. The defect amplitude of each set was varied in the ranges $0.06 \le \overline{\delta}_h \le 0.83, \ 0.18 \le \overline{\delta}_h \le 0.80, \ \text{and} \ 0.03 \le \overline{\delta}_h \le 0.73, \ \text{respectively}$. The results of the FEM simulations using the experimentally measured geometry are also presented and validated through direct comparison against the experimental results.

4.1. Load-carrying behavior

In Fig. 4, we present typical load-carrying curves, $\overline{p}(\overline{V}_{\rm im})$, for shells with a wide defect (fabricated using the nozzle $d_{\rm nozzle} = 4.8$ mm). In this plot, the critical buckling pressures associated with each curve are represented by the corresponding large symbols, which are normalized by the classic prediction for a prefect shell, $p_{\rm c}$. Naturally, the normalized critical buckling pressure is smaller than unity, due to the imperfection.

With a small defect amplitude (e.g., $\overline{\delta}_h = \{0.16, 0.37\}$), the normalized pressure loading \overline{p} increases with \overline{V}_{im} in the linear regime, until the peak value \overline{p}_1 (open symbols) is reached; this maximum value is the normalized critical buckling pressure of the shell. The shell buckles at \overline{p}_1 accompanied by an abrupt pressure drop. Past the onset of buckling, \overline{p} decreases gradually with \overline{V}_{im} , showing a significant loss in the load-bearing capacity of the shell, similarly to that of the imperfect shells with a dimple-like geometric defect (Hutchinson, 2016; Lee et al., 2016b).

When the defect amplitude is increased further (e.g., $\bar{\delta}_h = 0.51$), the load-carrying behavior is qualitatively different; the $\overline{p}(\overline{V}_{im})$ curve exhibits two peak pressure values in the loading path corresponding to two distinct buckling events. In this case, we denote the normalized critical pressures corresponding to the first and second buckling event by \overline{p}_1 (open symbols) and \overline{p}_2 (closed symbols), respectively. During the experiments, we observed that, at \overline{p}_1 , the shell buckled locally at the defect, while at \overline{p}_2 , the entire shell buckled globally. For this particular shell with $\overline{\delta}_h = 0.51$, we have $\overline{p}_1 = 0.24$ and $\overline{p}_2 = 0.22$. Since $\overline{p}_1 > \overline{p}_2$, the first event at \overline{p}_1 makes the shell unable to sustain any increasing pressure loading, although the second buckling at \overline{p}_2 was observed in the experiment with prescribed volume change. Again, if the shell were to be loaded with prescribed pressure, it would immediately collapse at \overline{p}_1 .

For even larger values of the defect amplitude (e.g., $\bar{b}_h = \{0.58, 0.68\}$), the qualitative features of the loading-carrying behavior change again. The value of the first buckling pressure is now found to be lower than that of the second event $(\bar{p}_1 < \bar{p}_2)$. Consequently, following the loading path, the shell first buckles at \bar{p}_1 with a small pressure drop, but the buckled shell is still able to carry the increasing pressure loading, until \bar{p}_2 is reached. In contrast to the previous case of $\bar{p}_1 > \bar{p}_2$, the two buckling events can be observed with either prescribed pressure or volume change.

4.2. Critical buckling pressure vs. the amplitude and the width of the defect

Thus far, from the loading curves measured in the experiments, we have found qualitatively distinct buckling behavior as the geometry of the through-thickness defect is varied. This motivates us to study the dependence of the critical buckling pressures on the defect amplitude and width in more detail. In Fig. 5, we plot the normalized buckling pressures, \bar{p}_1 (open symbols) and \bar{p}_2 (closed symbols), for the three sets of shells with increasing defect widths, as a function of normalized defect amplitude, $\bar{\delta}_h$.

With a relative narrow defect (Fig. 5A, fabricated using the small nozzle $d_{\text{nozzle}} = 3.5 \, \text{mm}$), the shell always exhibits a single buckling event. For these shells, increasing the defect amplitude changes the buckling behavior only quantitatively. In the range $0.06 \leq \overline{\delta}_h \leq 0.57$, the normalized buckling pressure, \overline{p}_1 , decreases from 0.82 to 0.26, and, for $\overline{\delta}_h \geq 0.57$, a nearly-flat plateau is reached at $\overline{p}_1 \approx 0.22$. This functional dependence is similar to that found previously for shells containing a dimple-like geometric defect (Hutchinson, 2016; Lee et al., 2016b). For large defect amplitudes, the fact that the plateau is not exactly constant (\overline{p}_1 decreases slightly with increasing $\overline{\delta}_h$) may be due to the unavoidable experimental interdependence during fabrication, even if small, between the setting of the defect amplitude (set mostly by the flow rate of the air jet) and the defect width (set mostly by the diameter of the nozzle).

For shells with a wider defect (Fig. 5B and 5C), the defect amplitude turns to qualitatively affect the critical buckling conditions beyond a threshold value for $\overline{\delta}_h$: $\overline{\delta}_h \geq 0.57$ in Fig. 5B and $\overline{\delta}_h \geq 0.48$ in Fig. 5C. In these ranges, two buckling events occur at \overline{p}_1 and \overline{p}_2 : \overline{p}_1 decreases monotonically with increasing $\overline{\delta}_h$, showing its sensitivity to the defect, while \overline{p}_2 follows a nearly-flat plateau.

Comparing the buckling pressures of the three sets of shells (Fig. 5), the plateau is lowered with increasing defect width. \overline{p}_1 reaches a plateau for the relatively narrow defect (Fig. 5A) with only one buckling event, while the plateaus for the two relatively wider defects are constructed by \overline{p}_2 (Fig. 5B and 5C). For all three cases, in contrast to the behavior at large values of $\overline{\delta}_h$, the defects fabricated using different nozzles do not yield significant differences in buckling pressure for $\overline{\delta}_h < 0.5$. This might be because the defect width is not sufficiently changed by varying nozzle diameters for $\overline{\delta}_h < 0.5$, as can be observed in the thickness profiles presented in Fig. 3.

In parallel to the experimental tests, we also conducted FEM simulations using the actual geometry measured from the experimental specimens through optical profilometry for all of the shells that were fabricated. These FEM results for the buckling pressures presented in Fig. 5 (\times and * symbols joined by dashed lines) show an excellent quantitative agreement with the experiments. This juxtaposition of FEM and experimental results validates that our numerical analysis can capture the buckling behavior of shells containing a through-thickness defect.

Before proceeding with presenting the more detailed results from the FEM and the theoretical analysis, we provide a partial summary of the findings uncovered thus far. We have developed an experimental protocol to fabricate shells containing a through-thickness defect, whose amplitude and width can be customized, and measured their critical buckling loads under pressure loading. The buckling behavior can be qualitatively different depending on the width of the defect. For narrow defects, the buckling behavior is 'classic'; the critical pressure decreases with defect amplitude, as a signature of the sensitivity to imperfections. For wider defects, above a threshold defect amplitude, two buckling events can occur: (i) local buckling at the defect and (ii) global buckling of the shell.

5. Buckling of shells containing a generalized defect profile: FEM and axisymmetric shell theory

Given the validity of our numerical analysis established in the previous section, we now focus on FEM to perform a systematic and extensive exploration of the parameter space that would be difficult (or cumbersome) to perform in the

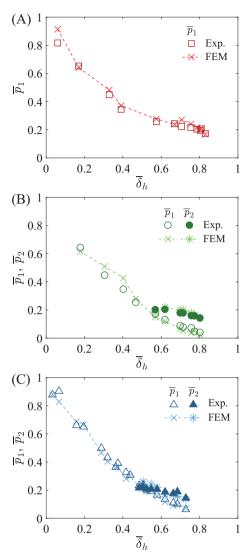


Fig. 5. Critical buckling conditions of the fabricated hemispherical shells containing a through-thickness defect. Normalized critical buckling pressures, \bar{p}_1 and \bar{p}_2 , versus normalized defect amplitude, $\bar{\delta}_h$, measured in the experiments and predicted by FEM. The defects are fabricated using air nozzles with different diameters: (A) $d_{\text{nozzle}} = 3.5 \, \text{mm}$, (B) $d_{\text{nozzle}} = 4.0 \, \text{mm}$, and (C) $d_{\text{nozzle}} = 4.8 \, \text{mm}$.

experiments. In addition, we shall contrast the FEM results to axisymmetric shell theory. For the sake of generality, we will consider the generalized through-thickness defect profile of Eq. (3). Specifically we will:

- (i) characterize various buckling regimes and quantify the associated critical buckling conditions of the system by independently varying the normalized defect amplitude $\overline{\delta}_h$ and half angular width $\overline{\beta}_h$ (Sections 5.1–5.4);
- (ii) investigate the influence of the width of the transition region across the edge of the defect, set by $\overline{\gamma}_h$ (Section 5.5);
- (iii) establish that the results are independent of radius-to-thickness ratio of the shell, R/h_0 (Section 5.6).

5.1. Load-carrying behavior and the associated buckling regimes

Before exploring the broad parameter space, we start by considering two representative sets of defects with $\overline{\beta}_h=1.2$ (relatively narrow) and $\overline{\beta}_h=2.6$ (relatively wide), with different amplitudes. In Fig. 6A, for the narrow defect case ($\overline{\beta}_h=1.2$) we plot the normalized pressure, \overline{p} , as a function of, respectively, the normalized displacement at the pole, \overline{w} , and the normalized volume change, \overline{V} . The corresponding values of $\overline{\delta}_h$ (for the various curves) are provided in the legend of the figure. With a small defect amplitude (e.g., $\overline{\delta}_h=0.1$), the shell initially deforms in the linear regime, where \overline{w} and \overline{V} increase linearly with \overline{p} . Buckling occurs when the normalized critical buckling pressure \overline{p}_1 (open symbols) is reached at $\overline{w}\approx 0.5$. Past this peak pressure, in the $\overline{p}(\overline{V})$ curve, both \overline{p} and \overline{V} decrease, closely following the upward branch. This indicates that the shell deformation occurs along the unstable branch and its strain energy decreases. Eventually, the shell stabilizes again at

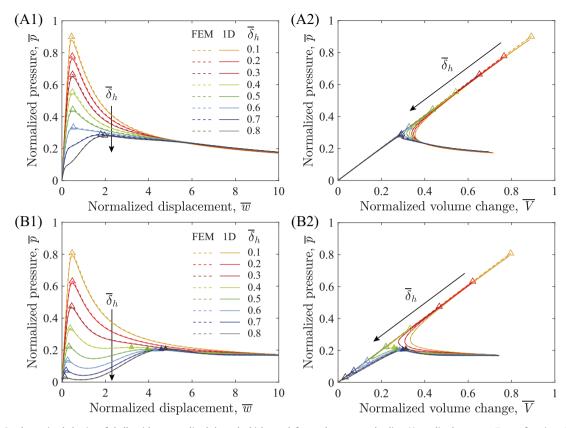


Fig. 6. Load-carrying behavior of shells with a generalized through-thickness defect under pressure loading. Normalized pressure, \overline{p} , as a function of (A1,B1) normalized pole deflection, \overline{w} , and (A2,B2) normalized volume change, \overline{V} , is presented for shells containing (A1,A2) a narrow defect $\overline{\beta}_h = 1.2$ or (B1,B2) a wide defect $\overline{\beta}_h = 2.6$: FEM (dashed lines) and axisymmetric shell theory (solid lines). The values of the normalized defect amplitude $\overline{\delta}_h$ for each $\overline{\beta}_h$ are provided in the legend. The critical buckling pressures associated with each curve are represented by the corresponding large symbols, open for \overline{p}_1 and closed for \overline{p}_2 .

the lower branch. The value of \overline{p}_1 decreases significantly as a function of the defect amplitude. Meanwhile, as the amplitude of the defect increases, \overline{p} becomes increasingly nonlinear with \overline{w} in the pre-buckling stage. For larger defect amplitudes $(\overline{\delta}_h \geq 0.7)$, the $\overline{p}(\overline{w})$ curve is remarkably nonlinear even in the early stage of loading, and the shell at the pole undergoes a large deformation $(\overline{w} \approx 2)$ prior to the onset of buckling. The corresponding critical buckling pressure, \overline{p}_1 , reaches a plateau and the unstable branch in the $\overline{p}(\overline{V})$ curve disappears.

We now turn to the shell containing a wider defect, as represented by the case with $\overline{\beta}_h=2.6$, whose $\overline{p}(\overline{w})$ and $\overline{p}(\overline{V})$ curves are plotted in Fig. 6B1 and 6B2, respectively. Overall, for $\overline{\delta}_h \leq 0.3$, the data shows similar trends to the previous narrow defect case ($\overline{\beta}_h=1.2$). However, the load-carrying behavior is qualitatively different when $\overline{\delta}_h \geq 0.4$; a second peak pressure, \overline{p}_2 , emerges in the loading path, following \overline{p}_1 . This first buckling event at \overline{p}_1 occurs in the early loading stage with a small deformation at the pole, which decreases gradually with increasing defect amplitude; e.g., from $\overline{w}=0.38$ at $\overline{\delta}_h=0.4$ to $\overline{w}=0.13$ at $\overline{\delta}_h=0.8$. Along the loading path, past \overline{p}_1 , the shell deforms in the unstable equilibrium state with decreasing trends for both \overline{p} and \overline{V} . In contrast with the above case for a narrow defect, as the shell stabilizes again at the lower branch, the pressure turns to increase, indicating that the buckled shell is still able to carry the increasing pressure loading. The shell at the pole continues to deform, with the development of a large dimple. By the time the second buckling condition $\overline{p}=\overline{p}_2$ is attained, this dimple has a depth of $\overline{w}\approx 4$. For shells exhibiting two buckling events, \overline{p}_1 decreases monotonically but \overline{p}_2 forms a plateau, leading to a swap in order of the pressure values associated with the two buckling events.

The two contrasting cases presented above suggest the presence of three robust buckling regimes, which we now seek to characterize further. To do so, we plot the normalized critical buckling pressures, \bar{p}_1 and \bar{p}_2 , and the associated normalized volume changes immediately prior to buckling, \bar{V}_1 and \bar{V}_2 , as a function of defect amplitude, for the two representative cases of a narrow defect $\bar{\beta}_h = 1.2$ (Fig. 7A) and a wide defect $\bar{\beta}_h = 2.6$ (Fig. 7B). In our description and interpretation of the buckling regimes, we will make use of the concept of *energy barrier* (Baumgarten and Kierfeld, 2018; 2019; Hutchinson and Thompson, 2017b; 2018; Knoche and Kierfeld, 2011; 2014; Marthelot et al., 2017), which the shell needs to overcome to attain the stable buckled state. This energy barrier, W, is the difference in the free energy of the shell between the prebuckling and the unstable buckled state, and can be written in dimensionless form as (Hutchinson and Thompson, 2018)

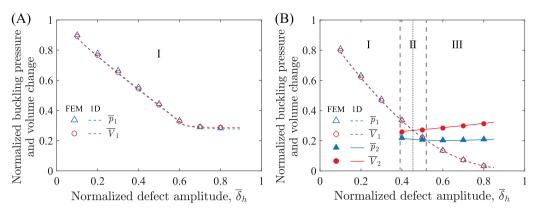


Fig. 7. Critical buckling pressures, \overline{p}_1 and \overline{p}_2 , and the associated normalized volume changes immediately prior to buckling, \overline{V}_1 and \overline{V}_2 , are plotted as a function of normalized defect amplitude $\overline{\delta}_h$, for (A) the narrow defect case $\overline{\beta}_h = 1.2$ and (B) the wide defect case $\overline{\beta}_h = 2.6$. The dashed black lines are the boundaries between different buckling regimes. The dotted black line represents the transition from $\overline{V}_1 \geq \overline{V}_2$ to $\overline{V}_1 < \overline{V}_2$ within Regime II.

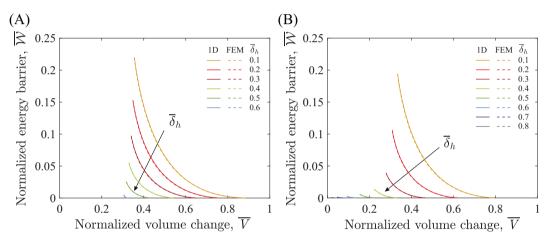


Fig. 8. Normalized energy barrier for the first buckling event, $\overline{\mathcal{W}}_h$, as a function of normalized prescribed volume change, $\overline{\mathcal{V}}_h$, for shells with (A) the narrow defect $\overline{\beta}_h = 1.2$ or (B) the wide defect $\overline{\beta}_h = 2.6$. The normalized defect amplitude for each $\overline{\beta}_h$ is varied $\overline{\delta}_h = \{0.1, 0.2, 0.3, 0.4, 0.5, 0.6, 0.7, 0.8\}$, as detailed in the legend.

$$\overline{W} = \frac{2(1-\nu)\sqrt{1-\nu^2}}{\sqrt{3}p_{\rm c}V_{\rm c}(h_{\rm o}/R)}W. \tag{11}$$

Next, we describe the features of the three buckling regimes in more detail.

Regime I: In this regime, only one buckling event occurs in the loading path at $\overline{p}_1(\overline{V}_1)$; \overline{p}_1 is the loading strength of the shell. With prescribed volume change, we now plot the normalized energy barrier, \overline{W} , as a function of the volume change, \overline{V} , in Fig. 8A for shells with a narrow defect ($\overline{\beta}_h = 1.2$), and, similarly, in Fig. 8B for shells with a wide defect ($\overline{\beta}_h = 2.6$). When there is an energy barrier, the shell snaps at \overline{p}_1 and jumps to the lower stable post-buckled state, inducing a pressure drop. At the same value of prescribed volume change, the energy barrier is lower for a larger defect amplitude, thus the buckling event is less catastrophic. In the absence of an energy barrier, no unstable branch exists in the loading path and the shell buckles with smooth changes in pressure and volume (rather than through snap-through buckling).

Regime II: In this regime, two buckling events occur sequentially along the loading path, first $\overline{p}_1(\overline{V}_1)$ and then $\overline{p}_2(\overline{V}_2)$, with $\overline{p}_1 \geq \overline{p}_2$. Under prescribed volume change, if $\overline{V}_1 \geq \overline{V}_2$, after the first buckling at $\overline{p}_1(\overline{V}_1)$, the second buckling will not occur. Conversely, $\overline{V}_1 < \overline{V}_2$ ensures that, after the first buckling event, \overline{V}_1 will reach \overline{V}_2 and trigger the second buckling event, as we have also observed in the experiments reported in Fig. 4 for $\overline{\delta}_h = 0.51$. To determine the type of buckling, we again examine the energy barrier under prescribed volume change and plot the results in Fig. 8B. We find that the shell undergoes snap-through buckling at \overline{p}_1 and the associated energy barrier decreases for increasing defect amplitude. Since there is no energy barrier required for the buckling at \overline{p}_2 , the shell buckles without snapping.

Regime III: In this regime, two buckling events occur in the loading path at $\overline{p}_1(\overline{V}_1)$ and $\overline{p}_2(\overline{V}_2)$, in a sequence with $\overline{p}_1 < \overline{p}_2$ and $\overline{V}_1 < \overline{V}_2$. The types of buckling at $\overline{p}_1(\overline{V}_1)$ and $\overline{p}_2(\overline{V}_2)$ are the same as those in Regime II. Given that $\overline{p}_1 < \overline{p}_2$ and $\overline{V}_1 < \overline{V}_2$ is a signature of this regime, past the initial snap-through buckling at $\overline{p}_1(\overline{V}_1)$, the shell carries the increasing pressure

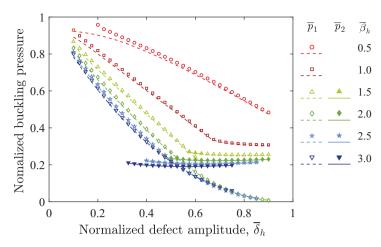


Fig. 9. Normalized critical buckling pressures, \bar{p}_1 and \bar{p}_2 , for shells with a generalized through-thickness defect versus normalized defect amplitude, $\bar{\delta}_h$, through FEM (symbols) and axisymmetric shell theory (lines). The normalized half angular width of the defect is varied $\bar{\beta}_h = \{0.5, 1.0, 1.5, 2.0, 2.5, 3.0\}$, as detailed in the legend.

loading until $\overline{p}_2(\overline{V}_2)$, for both pressure and volume controlled experiments. These two buckling events can be realized and observed in experiments, as we have shown in Fig. 4 for $\overline{\delta}_h = \{0.58, 0.68\}$.

Throughout all the panels in Figs. 6–8, results from the axisymmetric shell theory introduced in Section 3.3 were also reported. The excellent agreement found in all quantities investigated serves as a successful cross verification of the FEM and the 1D theoretical analysis. We recall that the generalized pure through-thickness defect investigated in this section is geometrically different than the imperfections that we were able to fabricate in the experimental samples. As such, at this stage, we cannot establish a quantitative comparison between the FEM/1D-theory results discussed in this section and the experiments (Section 4). Nonetheless, the three regimes for the buckling behavior that we have uncovered are, indeed, also observed in the experiments, which are unique for shells with a through-thickness defect. This finding indicates that, for the experimentally fabricated defects, the component in thickness governs the major features of shell buckling, despite their slightly non-spherical geometry of the middle surface.

5.2. Critical buckling pressures

Above, we showed that the critical buckling pressures, \overline{p}_1 and \overline{p}_2 , determine the loading strength and buckling regime of a shell. We proceed by performing a more thorough exploration of the parameter space to understand how the defect geometry dictates the critical buckling pressures. We shall focus exclusively on FEM and axisymmetric shell theory in the following range of parameters $\overline{\delta}_h \in [0.1, 0.9] \times \overline{\beta}_h \in [0.5, 4]$, while fixing $\overline{\gamma}_h = 0.4$ and $R/h_0 = 100$. The effect of varying $\overline{\gamma}_h$ and R/h_0 will be discussed in Sections 5.5 and 5.6.

In Fig. 9, we plot \overline{p}_1 and \overline{p}_2 as a function of the defect amplitude for different values of $\overline{\beta}_h$. We can then provide some additional insight on the mechanism underlying the buckling regimes, in terms of local buckling at the defect and global buckling of the entire shell. The critical pressure \overline{p}_1 shows different trends against the defect amplitude $\overline{\delta}_h$ in different regimes. In Regime I, represented by the absence of \overline{p}_2 , the local and global buckling are coupled and triggered synchronously at \overline{p}_1 , which decreases with increasing $\overline{\delta}_h$ and, eventually reaches a plateau (except for the special case of a very narrow defect with $\overline{\beta}_h = 0.5$). For these cases, \overline{p}_1 also decreases with increasing half angular width of the defect $\overline{\beta}_h$. By contrast, in Regimes II and III, the large defect decouples the local and global buckling. In this case, the local buckling is governed by the stiffness of the region of reduced thickness, therefore its critical buckling pressure \overline{p}_1 decreases monotonically with the thinning of this local region, but becomes insensitive to the defect width. While, the global buckling at \overline{p}_2 is governed by the stiffness of the constant-thickness region of the shell, exhibiting a plateau that is independent of the defect amplitude $\overline{\delta}_h$ but decreases with increasing $\overline{\beta}_h$.

5.3. Phase diagram for the buckling regimes

In Fig. 10, we construct a phase diagram in the $(\overline{\delta}_h, \overline{\beta}_h)$ parameter space for the three buckling regimes based on the relationship between \overline{p}_1 and \overline{p}_2 . The individual data points were computed through FEM, and the phase boundaries that separate the various regimes were determined using the axisymmetric shell theory, in excellent agreement with FEM.

For narrow defects with $\overline{\beta}_h < 1.25$, only Regime I is observed. Energy barrier is absent in Regime I for defect width $\overline{\beta}_h \ge 1.25$ and amplitude $\overline{\delta}_h \ge 0.65$ (closed red circles). Regimes II and III occur for $\overline{\beta}_h \ge 1.25$ with the reducing of the region of stability for Regime I. These imperfections are sufficiently wide to ensure the decoupling of the local buckling at the defect and global buckling of the entire shell. As discussed earlier, the defect amplitude plays a role in the relationship between \overline{p}_1

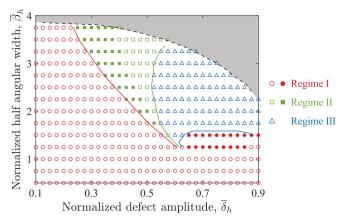


Fig. 10. Phase diagram of the buckling regimes for shells with a generalized through-thickness defect in the $(\overline{\delta}_h, \overline{\beta}_h)$ parameter space, $\overline{\delta}_h \in [0.1, 0.9]$ and $\overline{\beta}_h \in [0.5, 4]$. The symbols and boundary lines are the results from FEM and axisymmetric shell theory, respectively. In Regime I, buckling with or without an energy barrier under prescribed volume change is represented by the open or closed circles, respectively. In Regime II, shells with $\overline{V}_1 < \overline{V}_2$, whose second buckling event can be observed under prescribed volume change, are represented by open squares, while shells with $\overline{V}_1 \geq \overline{V}_2$ are represented by closed squares. Non-axisymmetric buckling occurs for shells in the gray region.

and \overline{p}_2 , thereby dictating the phase boundary between Regimes II and III. We note that shells in the gray region experience non-axisymmetric buckling due to the extremely thin and wide local region (Paulose and Nelson, 2013), which was also observed in our full 3D FEM simulations, but this regime is beyond the scope of this study.

This phase diagram quantifies the role of the geometry of a through-thickness defect in dictating the buckling regimes of our shells and further attests the ability of the axisymmetric (1D) shell theory to quantitatively describe the system, and capture the transition regions across the various buckling regimes.

5.4. Knockdown factor

We now focus on the knockdown factors, as defined in Eq. (2), for our imperfect shells containing a through-thickness defect. For shells buckling in Regime I, \overline{p}_1 is the loading strength normalized by the classic prediction. Thus, the associated knockdown factor is simply $\kappa_d = \overline{p}_1$. For shells buckling in Regimes II and III, the largest buckling pressure among \overline{p}_1 and \overline{p}_2 sets the loading strength of the shell, such that the corresponding knockdown factor is

$$\kappa_{\rm d} = \max\{\overline{p}_1, \overline{p}_2\}.$$
 (12)

According to the classification of the three buckling regimes that we introduced in Section 5.1, we have $\kappa_d = \overline{p}_1$ for Regimes I and II, and $\kappa_d = \overline{p}_2$ for Regime III.

In Fig. 11A, we now plot the knockdown factors for different defect amplitudes and widths. The corresponding contour map in the $(\overline{\delta}_h, \overline{\beta}_h)$ parameter space is presented in Fig. 11B.

From the data presented in both Fig. 11A and 11B, we find a clear plateau of knockdown factor for shells with a wide or/and deep defect. We determine the onset of the plateau by

$$\left| \frac{\mathrm{d}\kappa_{\mathrm{d}}}{\mathrm{d}\delta_{\mathrm{h}}} \right| \le \xi \,, \tag{13}$$

where ξ is a threshold chosen to be $\xi=0.2$. The onset of the plateau is represented by the open circle on each line in Fig. 11A and the corresponding defect amplitude at this onset, $\overline{\delta}_p$, is plotted as a function of $\overline{\beta}_h$ in Fig. 12B (solid line with circles). Below $\overline{\beta}_h=0.75$, for a very narrow defect, κ_d does not reach a plateau in the ranges of parameters explored. A plateau does develop for $\overline{\beta}_h \geq 0.75$, and its onset gradually moves towards lower values of $\overline{\delta}_h$ (Fig. 12B). $\overline{\delta}_p$ reaches the minimum of 0.52 at $\overline{\beta}_h=2.5$. The plateau knockdown factor κ_p shown in Fig. 12A (solid line with circles) is the average of κ_d over the extent of the plateau, which is monotonously decreasing with increasing defect width from $\kappa_p=0.36$ at $\overline{\beta}_h=0.75$ down to $\kappa_p=0.19$ at $\overline{\beta}_h=3.25$.

5.5. Effect of the width of the transition region across the edge of the defect

In this section, we will discuss the effect of the geometric parameter $\overline{\gamma}_h$ on both the critical buckling pressures and the plateau of knockdown factor. This parameter $\overline{\gamma}_h$ sets the width of the transition region across the edge of the defect, as shown in the inset of Fig. 13. The smaller $\overline{\gamma}_h$, the steeper the transition. However, the varied $\overline{\gamma}_h = \{0.2, 0.4, 0.6\}$ (see legend of the plot) just slightly affects the critical buckling pressures, as shown in Fig. 13. We also plot the corresponding plateau

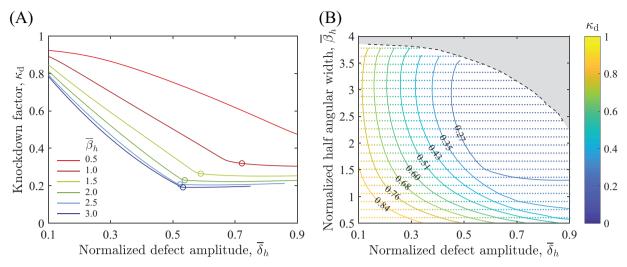


Fig. 11. Knockdown factors of shells with a generalized through-thickness defect. (A) knockdown factor, κ_d , versus normalized defect amplitude, $\overline{\delta}_h$. The normalized half angular width of the defect is varied $\overline{\beta}_h = \{0.5, 1.0, 1.5, 2.0, 2.5, 3.0\}$, as detailed in the legend. The open circle on each line represents the onset of the plateau. (b) Contour plot of the knockdown factor for different values of the normalized defect amplitude, $\overline{\delta}_h \in [0.1, 0.9]$, and normalized half angular width, $\overline{\beta}_h \in [0.5, 4]$. Non-axisymmetric buckling occurs for shells in the gray region.

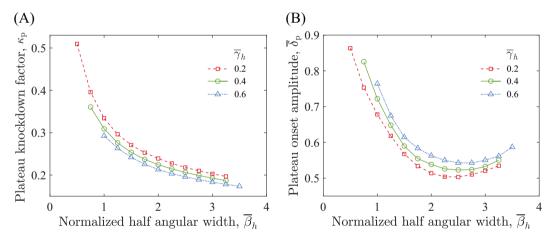


Fig. 12. (A) Normalized pressure level of the plateau, κ_p , versus normalized half angular width of the defect, $\overline{\beta}_h$. (B) Normalized defect amplitude at onset of the plateau, $\overline{\delta}_p$, versus $\overline{\beta}_h$. The width of the transition region across the edge of the defect is varied by changing $\overline{\gamma}_h = \{0.2, 0.4, 0.6\}$.

knockdown factor, κ_p , in Fig. 12A. We can find that, κ_p slightly decreases with increasing $\overline{\gamma}_h$, since the wide transition region effectively enlarges the defect width, but the effect is small. On the other hand, the region of reduced thickness close to the edge of the defect becomes thicker with increasing $\overline{\gamma}_h$. As a result, the critical buckling pressure for local buckling, \overline{p}_1 for Regimes II and III, is increased. An increasing $\overline{\gamma}_h$ also pushes the onset of plateau moving towards higher values of defect amplitude, as shown in Fig. 12B.

5.6. Independence of radius-to-thickness ratio of the shell

Thus far, the radius-to-thickness ratio was fixed at $R/h_0=100$, but we will now establish that the results that we have obtained are independent of this value. For this purpose, we obtain the critical buckling pressures for $R/h_0=\{50,100,200\}$ (see legend of the plot) as a function of the normalized defect amplitude through the developed axisymmetric shell theory, as shown in Fig. 14. We also vary the normalized half angular width of the defect $\overline{\beta}_h=\{1.0,1.5,2.5\}$. It can be seen that, \overline{p}_1 and \overline{p}_2 versus $\overline{\delta}_h$ are not affected by R/h_0 , while fixing $\overline{\beta}_h$ and $\overline{\gamma}_h$. We can conclude that the buckling behavior of the shell is entirely governed by the three geometric parameters of the through-thickness defect, $\overline{\delta}_h$, $\overline{\beta}_h$ and $\overline{\gamma}_h$, thereby ensuring that the results presented in the paper are applicable to thin spherical shells.

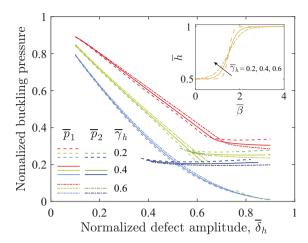


Fig. 13. Normalized critical buckling pressures, \overline{p}_1 and \overline{p}_2 , versus normalized defect amplitude, $\overline{\delta}_h$, for different widths of the transition region across the edge of the defect: $\overline{\gamma}_h = 0.2$ (dashed lines), $\overline{\gamma}_h = 0.4$ (solid lines), and $\overline{\gamma}_h = 0.6$ (dash-dotted lines), through the axisymmetric shell theory. The normalized half angular width of the defect is varied: $\underline{\beta}_h = 1.0$ (red lines), $\overline{\beta}_h = 1.5$ (green lines), and $\overline{\beta}_h = 2.5$ (blue lines). Inset: Thickness profiles of shells with varied $\overline{\gamma}_h = \{0.2, 0.4, 0.6\}$ for $\overline{\delta}_h = 0.5$ and $\overline{\beta}_h = 1.5$ (For interpretation of the references to color in this figure legend, the reader is referred to the web version of this article.).

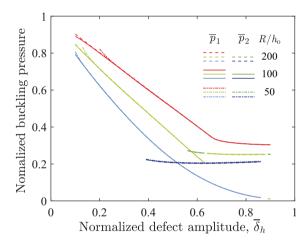


Fig. 14. Normalized critical buckling pressures, \overline{p}_1 and \overline{p}_2 , versus normalized defect amplitude, $\overline{\delta}_h$, for shells with different radius-to-thickness ratio: $R/h_0 = 200$ (dashed lines), $R/h_0 = 100$ (solid lines), and $R/h_0 = 50$ (dash-dotted lines), through the axisymmetric shell theory. The normalized half angular width of the defect is varied: $\overline{\rho}_h = 1.0$ (red lines), $\overline{\rho}_h = 1.5$ (green lines), and $\overline{\rho}_h = 2.5$ (blue lines). The width of the transition region across the edge of the defect is fixed at $\overline{\gamma}_h = 0.4$ (For interpretation of the references to color in this figure legend, the reader is referred to the web version of this article.).

6. Conclusion

We have studied the buckling of pressurized spherical shells containing a through-thickness defect, and identified three buckling regimes. In Regime I, the shell exhibits only one buckling event and its load-carrying behavior is similar to a constant thickness shell containing a local geometric defect. The buckling behavior of Regimes II and III is qualitatively different, showing two buckling events. In Regime II, local buckling at the defect is the first buckling event that has a high critical pressure, which will simultaneously trigger global buckling under prescribed pressure. In Regime III, the two buckling events can always be observed with either prescribed pressure or volume change. We summarized these buckling regimes in a phase diagram, representing the dependence of the phase boundaries on the geometry of the defect. We also systematically analyzed the critical buckling conditions affected by the defect and the underlying mechanism involving local and global buckling, which are dictated by the three geometric parameters of the defect, independent of the radius-to-thickness ratio of the shell. Buckling behavior similar to Regimes I and III of spherical shells has also been observed in cylindrical shells under axial compression (Gerasimidis et al., 2018; Haynie et al., 2012; Kriegesmann et al., 2012). In these cases with cylindrical shells, local buckling is triggered by a geometric imperfection or a lateral force probe, which always has a value of critical load/shortening lower than that corresponding to global buckling, at least in the explored parameter space. The buckling of cylindrical shells in the presence of thickness variation is yet unexplored and would be an interesting topic for future studies.

Although the buckling strength is reduced by the defect, the shell along the plateau has a deterministic critical buckling load, which can be predicted. This indicates that, a precisely engineered through-thickness defect could be used to accurately control the onset of buckling, when used for functionality. For example, under prescribed pressure, the through-thickness defect induces two buckling events, with one stable post-buckled state for the shells in Regime III. The two bucking events can be triggered in sequence by adjusting the external pressure. This could be used as an actuator in soft robotics to execute sequential actions, or in drug delivery as a capsule to release drugs at two different speeds through its break-downs by local and global buckling.

The agreement between experiments and FEM simulations presented in this study further concretes the finding that numerics can give an accurate prediction on critical buckling loads, as long as the imperfections are properly considered. Recent developments in imaging techniques, *e.g.*, optical profilometry or x-ray computed tomography, are making it possible to access the 3D geometry of structures. Further numerical simulations are required to make use of the measured full 3D geometric information of a specific structure and produce an accurate prediction for the buckling strength. This could be used to study the effect of a distribution of defects on the critical load, which represents a more realistic scenario. We hope our study will provide valuable physical insight for the engineering design of shell structures, and stimulate future studies on this topic.

Declaration of Competing Interest

The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.

Appendix A. Imposed volume change in the experiments

In Section 3.1.3, we noted that, in the experiments, the volume of the air expansion imposed by the syringe, $V_{\rm im}$, is not equal to the volume change of the shell, V. This difference is due to the compressibility of air, even in the pre-buckling regime. Next, we will detail the relationship between $V_{\rm im}$ and V.

We shall refer to the pressure differential between the outside and inside of the shell as $p = p_{\text{out}} - p_{\text{in}}$, and to the volume change of the shell as $V = -(V_{\text{shell}} - V_{\text{shell0}})$, where V_{shell} is the volume of the shell. Quantities with the subscript "0" refer to the reference state (p = 0). In our experiments, the shell specimen is tested by connecting it to a syringe via a thick-walled silicone tube. The syringe is used to expand the air in the system (volume V_{system}), which comprises the shell, syringe and connecting tube. As a result, the internal pressure in the system will decrease, thus, the shell is pressurized by the positive outside/inside pressure differential.

For the reference state ($V_{im} = 0$ and p = 0) the ideal gas law yields

$$p_{\rm in0}V_{\rm system0} = Nk_{\rm B}T\,, \tag{A.1}$$

where N is the number of gas molecules in the system, $k_{\rm B}$ is the Boltzmann constant, and T is the absolute temperature of the gas. During the expansion of the air inside the system, we have

$$(p_{\text{in0}} - p)(V_{\text{system0}} - V + V_{\text{im}}) = Nk_{\text{B}}T.$$
 (A.2)

Assuming an isothermal process, combining Eqs. (A.1) and (A.2) yields,

$$\frac{V_{\text{system0}}}{-V + V_{\text{im}}} = \frac{p_{\text{in0}}}{p} - 1. \tag{A.3}$$

For our experimental shells, classic shell theory (for a perfect spherical shell) predicts a critical buckling pressure of $p_c = 173 \, \text{Pa}$, which is significantly smaller than $p_{\text{in0}} = 1.01 \times 10^5 \, \text{Pa}$, such that $p_{\text{in0}}/p \gg 1$. Therefore, we obtain

$$\frac{V_{\text{system0}}}{-V + V_{\text{im}}} \approx \frac{p_{\text{in0}}}{p} , \qquad (A.4)$$

which can be rearranged to read

$$V_{\rm im} pprox rac{p}{p_{\rm in0}} V_{
m system0} + V$$
 . (A.5)

The above equation establishes the dependence between the volume change of the shell and the volume change imposed by the syringe. In the pre-buckling regime, for a perfect hemispherical shell, we have p = KV, where $K = Eh_0/\pi (1 - \nu)R^4$ is the stiffness of the shell (Hutchinson and Thompson, 2017a). Hence, we can rewrite Eq. (A.5) as

$$V_{\rm im} \approx \left(\frac{KV_{\rm system0}}{p_{\rm in0}} + 1\right)V$$
 (A.6)

Therefore, we find that $V_{\rm im}$ depends linearly on V, with $V_{\rm im} > V$. Moreover, the prefactor in Eq. (A.6), $KV_{\rm system0}/p_{\rm in0}+1$, depends on both the volume of the air inside the system and the shell stiffness. For our specific experimental conditions, we have $V_{\rm system0} = 5.4 \times 10^4 \, \rm mm^3$, $K = 0.48 \, \rm Pa/mm^3$, and $p_{\rm in0} = 1.01 \times 10^5 \, \rm Pa$, yielding $V_{\rm im} \approx 1.25 V$.

Supplementary material

Supplementary material associated with this article can be found, in the online version, at doi:10.1016/j.jmps.2020.103923

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